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ASSIGN REVEWER

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OXCART WEEKLY REVIEW

8 January - 25 January

	L. THE GLINE CHARLES CHARLES CONTROL OF CONTROL CONTRO	
25X1A	Fratt & Whitney the hydraulic pump problems connected with the J-58	25X1A
25X1A	b. On 9 and 10 January sttended the SAE pro- fessional meetings in Detroit, Michigan.	
25X1A	m An 33 Yoursens	25X1 <i>A</i>
25X1A	d. On 15 January Messrs. Kiefer, Parangosky and of Head-	25X1A
25X1A	quarters met with of ITEK and to discuss the linear correlator problems in connection with	25X1 <i>F</i> -
25X1A		
25X1A	f. Colonel Beerli,	
25X1A	familiarization course at Eurbank between 16 and 18 January.	25X1A
25X1A	trip report, OXC-2973, dated 22 January 1962, has been forwarded to DD/P.	
25X1A	g. Eugene Kiefer was and LAC between 16 and 18 January in connection with and the Kirkpatrick visit	25X1 <i>F</i>
25X1A		
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	ence the P&I	local fligh	C-23 arrived at test program. s mission.					
i. Bet	ween 15 and	19 January						
of were	IDY LAC wher	e they works	d with	on fixes f	or two			
problems in	connection w	ith the pare	chute evaluation	exercise.	The result			
(1) Firewel i	s reworking	the faulty links	ige system.				
(2) is reworking and strengthening the harnesses.								
perachute ter for the drop	sting progra s. It is pl	m which will anned that t	se plane, has be give us a capab ests will be res se linkage system	pility of 50 numed on 29	,000 feet January			
ğ _{ir} .			Inch de Parenten la					
<u>k. </u>	1 4		that in January k					
forwarded to		CLTD Labor	, URC-EYIO OL ZZ	: variumry, u	ere nach			
101 400 000 00	22/2							
1. Engi	ine test tim	e accumulate	d for the period	1 - 23 Jan	wary 1962:			
To	tal engine t	ine	73 hours					
Afterburner time			2 hours					
D-20 engine time			73 hours					
Hot inlet time			Ö					
Ho	t turbine ti	Re	1 hour					
) Four sea 113, XD-1,		tands are in ope	eration with	engines			

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Engine FX-116 (ejector) is mounting in No. 5 see level stand.

for inspection after test completion.

(2) Engine FX-115 (endurance) is mounting in the altitude stand.

(3) Engines FX-112, 114, and 118 have been returned to assembly

(4) Primary effort continues on turbine inlet temperature profile; secondary effort on hydraulic system and controls development.

OXC-2989 Copy 9 of 9 Page 3 (5) The favorable turbine inlet temperature profile recently established on engine FX-113 has proved inconsistent. 25X1A feels that this is a further indication of diffuser case flow instability, for which initial corrective hardware is now on test with FX-111 and XD-1. This will delay initiation of endurance testing well beyond the three weeks estimated on 4 January. (6) FX-114 teardown inspection has revealed good condition of bolted compressor rotor but damage to most turbine blades and vanes resulting from one burner can mounting bolt failure. (7) The first fully integrated single lever control system using non-production hardware completed 14 hours of non afterburning operation on engine FX-114 without instability. Two combinations of a similar fully integrated single lever system have completed 5 non afterburning hours on engine XD-2 without instability. XD-2, with the latter fully integrated but two lever system, has initiated afterburning operation without instability. The second lever is required to initiate afterburning without fulfilling rated turbine inlet temperature. (8) The first production main fuel and exhaust nozzle controls have been delivered to Florida. 2. Agenda for 26 January Development Branch meeting. 25X1A a. Further Discussion of Engine Status if Required b. Possible Rescheduling of A-12 Trainer - Mr. Kiefer. 25X1A e. Cover Story Status d. Change in Aircraft Engine Out Ferformance - Mr. Kiefer. 25X1A e. Report on the Kirkpatrick Visit - Mr. Perangosky. 1.1 Distribution: JOHN PARANGOEKY Copy #1 - DD/P C/DB/DPD 2 - AC/DPD 3 - EXO/DPD 4 - ASST C/DPD 5&6 - C/DB/DPD 7 - SA/TA/DPD 8 - DB/DPD9 - RI/DPD

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DC/DB/DPD/EFF:hmj (25 Jan 62)